

# SUPPLEMENTAL TYPE CERTIFICATE 10057054

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

## AIRBUS HELICOPTERS CANADA LIMITED

1100 GILMORE ROAD FORT ERIE ON L2A 5M9 CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Type Certificate Number: EASA.R.008** 

EASA.R.146

Type Certificate Holder: AIRBUS HELICOPTERS

Type: AS 350/EC 130

AS 355

Model: AS 350 B, AS 350 B1, AS 350 B2

AS 350 B3, AS 350 BA, AS 350 D

AS 355 E, AS 355 F, AS 355 F1

AS 355 F2, AS 355 N, AS 355 NP

Original STC Number: TCCA STC SH98-43

#### **Description of Design Change:**

Installation of Rear Crosstube Maintenance Steps (350-500604).

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 04 March 2016

Pier Giorgio COLOMBO

Medium Rotorcraft Section

Manager

10038530

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This STC is a validation of Canadian STC no. SH98-43.

## **EASA Certification Basis:**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

## **Associated Technical Documentation:**

Airbus Helicopters Canada Limited Master Drawing List MDL-98-002, Issue C, Transport Canada approved 09 December 2014

Airbus Helicopters Canada Limited Instructions for continued Airworthiness ICA-ECL-235, revision 1, Transport Canada accepted 09 December, 2014

or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/ Canada Bilateral Agreement.

## **Limitations/Conditions:**

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.