



**SUPPLEMENTAL TYPE CERTIFICATE**

**10016747, REV. 1**

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

**EUROCOPTER CANADA Ltd.**

**1100 GILMORE ROAD  
P.O. BOX 205  
FORT ERIE ON L2A 5M9  
CANADA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Type Certificate Number : EASA.R.008  
Original Type Certificate Number : EASA.R.146  
Type Certificate Holder : EUROCOPTER  
Type Design - Model : AS350B/BA/B1/B2/B3/D  
Type Design - Model : AS355E/F/F1/F2  
Original STC Number : TCCA STC SH96-31**

**Description of Design Change:**

Relocation of higher capacity battery to tail boom (mod. P/N 350-700324 and 355-700324)  
Revision 1 incorporates AS350B3 model configurations incorporating new major design modifications  
This STC is the validation of TCCA STC SH96-31 at issue 9.

**EASA Certification Basis:**

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.  
The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

**For the European Aviation Safety Agency,**

**Date of issue: 29 October 2013**

  
**Massimo MAZZOLETTI  
Certification Manager  
Rotorcraft, Balloons, Airships**

Note.  
The following numbers are listed on the certificate:  
EASA old Project Number: EASA.IM.R.S.01073, REV. 1

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**Associated Technical Documentation:**

Flight Manual Supplement - Battery Relocation AS350 / AS355 - doc. ref. FMS-ECL-155 Rev. 1  
Section 2 "Airworthiness Limitations" of Instructions for Continued Airworthiness - Battery Relocation AS350 / AS355 - doc. ref. ICA-ECL-125 Rev. 12

or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/ Canada Bilateral Agreement.

ECL Master Drawing List - Battery Relocation AS350 - doc. ref. MDL-04-003 Issue H (for AS350 models)

ECL Master Drawing List - Battery Relocation AS355 - doc. ref. MDL-96-001 Issue M (for AS355 models)

Instructions for Continued Airworthiness - Battery Relocation AS350 / AS355, excluding Section 2 "Airworthiness Limitations" - doc. ref. ICA ECL-125 Rev. 12

**Limitations/Conditions:**

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

The installation of this modification by third persons is subject to written permission of the approval holder.

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**Note:**

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