

3 Baseline Aircraft Definition

GENERAL

- · Energy absorbing fuselage
- Tail boom with fixed horizontal stabilizer and two endplates
- Upper deck with fittings for main gearbox, engines, hydraulic and cooling system
- Cowlings for main transmission and engines
- Skid-type landing gear with skid protectors, capable of taking ground-handling wheels
- · Long boarding steps, LH and RH
- · Cold weather kit
- · Built-in maintenance steps and grips
- Exterior painting (single color)

COCKPIT, CABIN AND CARGO COMPARTMENT

- One-level cabin and cargo compartment floor with integrated rails
- Glazed canopy
- Two hinged cockpit doors
- · Sliding window on pilot door
- · Map case in pilot's door
- Two wide passenger sliding doors with window of pushout type
- Two rear hinged clam-shell doors
- Longitudinally adjustable energy absorbing pilot seat with head rest and 4-point safety belt with automatic locking system; seat color grey or blue
- · Cabin boarding grips (LH and RH)
- · Interior paneling

- Flight controls (pilot side)
- Engine twist grip controls at pilot's collective pitch lever
- Instrument panel with extension on pilot's side and glare shield
- · Ram-air for cockpit
- · Electrical ventilating system for cockpit
- · Headset holder in the cockpit, rotatable
- · Portable fire extinguisher
- · Stowage net for first aid kit at the LH rear clam-shell door
- 1 flashlight (torch)
- Slant panel
- Center console
- Windscreen wiper for pilot and copilot
- Door open warning

BASIC INSTRUMENTATION

CENTRAL PANEL

- Central Panel Display System (CPDS) consisting of two LCD displays
 - Cautions and Advisories Display (CAD) with digital indication of:
 - Caution and advisory information
 - Fuel quantity indication
 - Vehicle and Engine Multifunction Display (VEMD) with digital indication of:
 - Engine parameters (engine oil pressure, engine oil temperature)
 - \bullet FLI (First Limit Indicator) for TQ, TOT, $\Delta N1$ as analogue display
 - Main gear box parameters (oil pressure, oil temperature)
 - Dual amperemeter for generator; ampere meter for battery
 - Dual voltmeter
 - Outside Air Temperature (OAT)
 - Mast moment indication

- Back-up conventional instruments (2 in)
 - Clock
 - Stand-by-horizon
 - Triple (rotor and engines) RPM-indicator
 - Air speed indicator
 - All speed
 Altimeter
- Warning unit
 - Engine fire warning with fuel emergency shut-off
 - Warning lights
 - Aural warning (for each warning, rotor RPM, fire warning)
 - Fire extinguishing system warning
 - Master caution light
- Main switch panel
 - DC power control
 - VARTOMS control
- Start switches
- Magnetic compass

AIR DATA

- Dual pitot static system (electrically heated pitot tube and static port)
- 2 ADC MEGHAS sensors

POWER PLANT

- Two TURBOMECA ARRIEL 1E2 turbine engines, complete with starting, fuel supply and control systems
- Crash resistant fuel system
- Two independent oil cooling and lubrication systems of the engines
- Fire detection and extinguishing system
- VARTOMS (Variable Rotor speed and Torque Matching System)
- Overspeed control
- Cycle counter



TRANSMISSION SYSTEM

- Main transmission including an independent redundant lubrication system and monitoring sensors
- · Rotor brake system

• Tail rotor transmission and intermediate transmission with splash lubrication, magnetic plug and oil level sight gauge

ROTOR AND FLIGHT CONTROLS

- Hingeless main rotor (System Bölkow) with 4 glass and carbon fiber reinforced blades with erosion protection strip
- Semi-rigid tail rotor with 2 twisted glass fiber reinforced blades of new technology with erosion protection strip
- · Basic provisions for an easy integration of a balancing
- Dual hydraulic boost system for cyclic and collective blade control of the main rotor
- Single hydraulic boost system for yaw control
- Stability Augmentation System (SAS) for tail rotor
- · Main and tail rotor blade tip painting (yellow)

ELECTRICAL INSTALLATION

- Power generation system:
 - Two starter/generators (2 x 200 A, 28 VDC)
 - Nickel-Cadmium battery, (24 V, 27 Ah), rear installation
 - External power connector (STANAG 3302)
- Power distribution system:
 - Two primary busbars
 - Two essential busbars
 - Two shedding busbars
 - Two non-essential busbars (50 A) for optional equipment only
 - · Battery bus
 - One utility receptacle in cargo comp. (28 VDC, 20 A)

- · Lighting:
 - Anti-collision warning light (red flashing)
 - Fixed landing light (250 W)
 - Three position lights (red, green, white)
 - · Adjustable instrument lighting
 - · One utility light in the cockpit
 - Lights in the cabin and cargo compartment
 - · Boarding illumination
 - · Emergency lights
- Radio:
 - Two radio master switches

GROUND HANDLING KIT^a

- Two ground-handling wheels
- Basic aircraft covers (short time)
- Keys for cockpit doors, cabin doors, ba compartment doors and tank flap (one-key system) doors, baggage
- · Battery key
- · Lifting points
- · Compass compensation key
- · Fuel drain device
- a. Weight not included in the standard helicopter empty weight

DOCUMENTATION (in English)

- One Flight Manual^{a b} (on paper)
- One Pilots Checklist^C (on paper)
- One Master Minimum Equipment List (MMEL)^a (on paper)
- One Logbook (on paper, CD-ROM on demand)
- One Historical Record (on paper, CD-ROM on demand)
- Technical Documentation^{ad} incl. AMM, SDS, WDM, IPC, MSM, CECG, SRM online via KEYCOPTER[®] portal
- Service Bulletin Catalogue (SB) online via T.I.P.I
- One List of Applicable Publications (LOAP)^a online via KEYCOPTER[®] portal
- One Avionics Manual^e (for avionics installed by Airbus Helicopters) (on CD-ROM)
- CD-ROM ECMM^C (Electronic Maintenance Manuals) for vendor manuals
- One Engine Documentation^f (on paper or CD-ROM), furnished by supplier, including:
- Illustrated Parts Catalogue (IPC)
- a. Revision service included as long as the aircraft is operational
- b. One Flight Manual included in the standard helicopter empty weight
- c. Revision service for 3 years d. Customized AMM, SDS, WDM and IPC versions available on request
- e. Customized documentation f. Revision service for 5 years

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