

# SUPPLEMENTAL TYPE CERTIFICATE 10041637

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 1702/2003 to

# **EUROCOPTER CANADA Ltd.**

1100 GILMORE ROAD P.O. BOX 205 FORT ERIE ON L2A 5M9 CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Product TC Number: EASA.R.008

TC Holder: EUROCOPTER

Model: EC130B4

Original STC Number: TCCA STC SH12-10

**Description of Design Change:** 

Door handle installation (P/N 120-200014)
This STC is the validation of TCCA STC SH12-10 (issue date -February 10, 2012)

### **EASA Certification Basis:**

The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 03.10.2012

Massimo MAZZOLETTI
Certification Manager
Rotorcraft, Balloons, Airships

Note:

The following numbers are listed on the certificate: EASA current Project Number: 0010019482-001

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## **European Aviation Safety Agency**



### **Associated Technical Documentation:**

Definition and installation:

ECL Master Drawing List - doc. ref. MDL-10-015 Issue B

Installation Procedure - RH Door Handle Installation - EC130B4 # doc. ref. IP-ECL-127 Rev. 2 Inspection and Maintenance:

Instructions for Continued Airworthiness - RH Door Handle Installation - EC130B4 - doc. ref.

ICA-ECL-204 Rev. 0

#### Limitations:

None

#### Conditions:

Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

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