

SUPPLEMENTAL TYPE CERTIFICATE

10057612

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

AIRBUS HELICOPTERS CANADA LIMITED

1100 GILMORE ROAD
FORT ERIE ON L2A 5M9
CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.R.008

Type Certificate Holder: AIRBUS HELICOPTERS

Type: AS 350/EC 130

Model: AS 350 B

AS 350 B1

AS 350 B2

AS 350 B3

AS 350 BA

AS 350 BB

AS 350 D

EC 130 B4

EC 130 T2

Original STC Number: TCCA SH16-11 ISSUE 1

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 30 March 2016



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**Medium Rotorcraft Section
Manager**

10042933

SUPPLEMENTAL TYPE CERTIFICATE - 10057612 - AIRBUS HELICOPTERS CANADA LIMITED - 301635



Description of Design Change:

Provision for ENGINE AIR INTAKE COVER installation, P/N 350-700924.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.
The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

Master Drawing List MDL-15-006 Issue C, 1st March 2016.
Instructions for Continued Airworthiness ICA-AHCA-296 revision 0, 1st March 2016.
Compliance Program CP-0401 Issue A, 1st March 2016.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -