



European Aviation Safety Agency

SUPPLEMENTAL TYPE CERTIFICATE

10047881

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

EUROCOPTER CANADA Ltd.

**1100 GILMORE ROAD
P.O. BOX 250
FORT ERIE ON L2A 5M9
CANADA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number : 1. EASA.R.008
Original Type Certificate Number : 2. EASA.R.146
Type Certificate Holder : FRANCE
Type Design - Model : 1. AS350 B/D/B1/B2/BA
Type Design - Model : 2. AS 355 E/F/F1/F2/N/NP
Original STC Number : TCCA STC SH97-33 ISSUE 3

Description of Design Change:

Rear Cargo Door Hinge and Latch Modification

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 27 January 2014


Massimo MAZZOLETTI
Certification Manager
Rotorcraft, Balloons, Airships

Note:
The following numbers are listed on the certificate:
EASA current Project Number: 0010028441-001

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Associated Technical Documentation:

Eurocopter Canada Ltd Master Drawing List MDL-98-008 Issue C

Eurocopter Canada Ltd Instructions for Continued Airworthiness ICA-ECL-207, rev. 2

or later revisions of the above listed documents accepted by EASA in accordance with the Technical Implementation Procedures of EU/ Canada Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

Note:

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